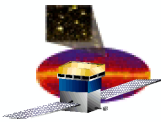


# GLAST Large Area Telescope:

## Technical Management and Open Technical Issues

Lowell A. Klaisner  
SLAC  
Chief Engineer

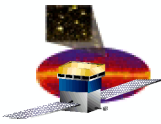
[klaisner@slac.stanford.edu](mailto:klaisner@slac.stanford.edu)  
650-926-2726



# Outline

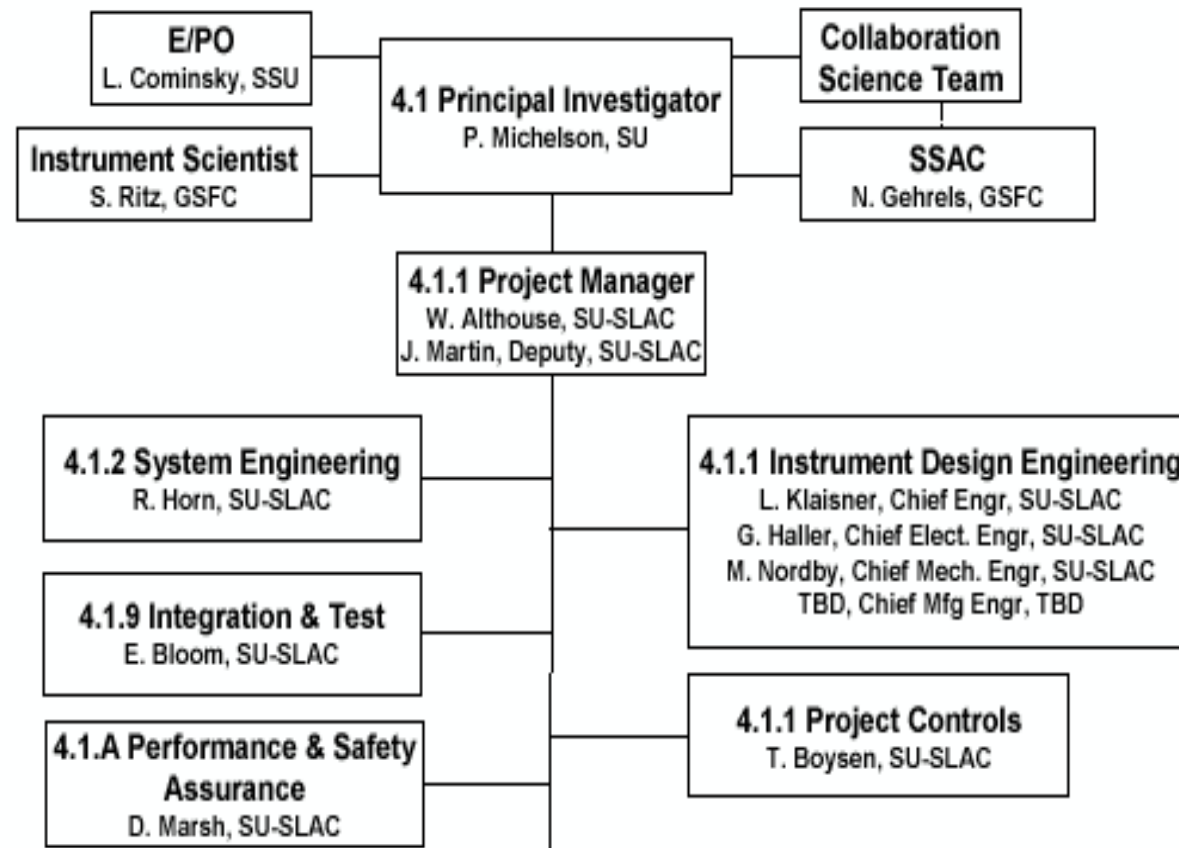
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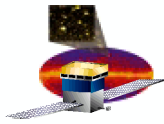
- **Technical Management Organization**
- **Open Technical Issues**
  - Tracker Bottom Tray
  - Calorimeter photodiodes
  - Application Specific Integrated Circuits (ASICs)
  - Other issues
- **Scope of the design effort**
- **Engineering Model Deliverables**
- **Schedule**
- **Spacecraft**
- **Summary**



# Tech Management – Project Office – 1 of 2

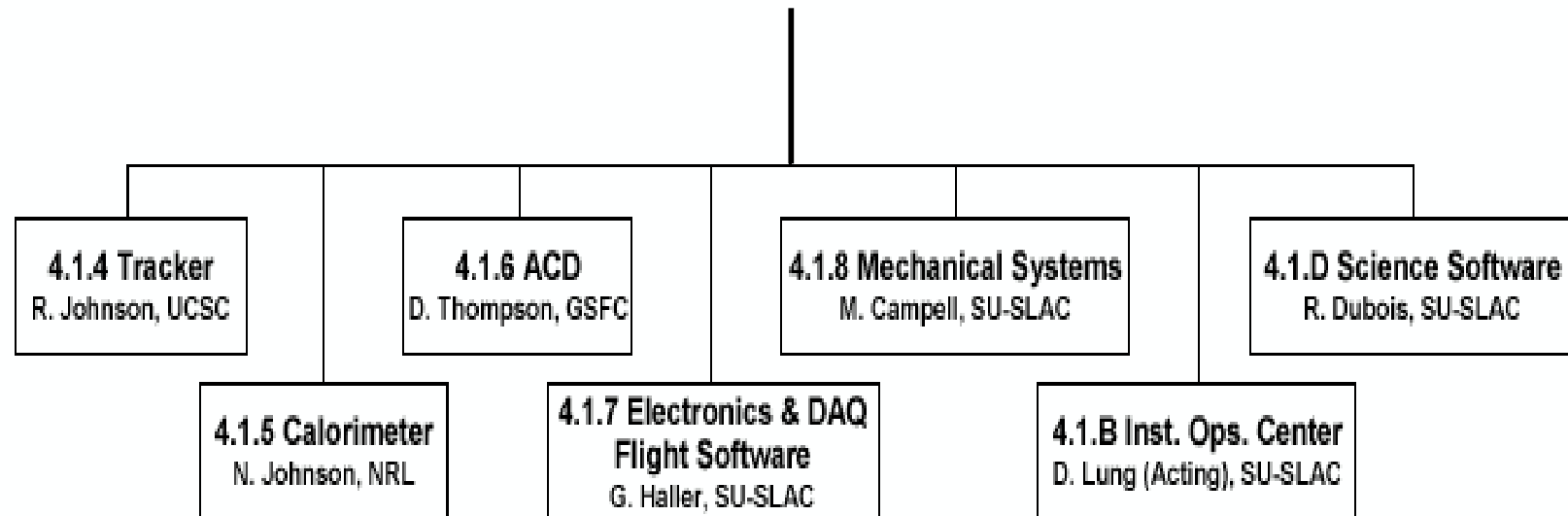
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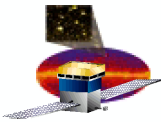




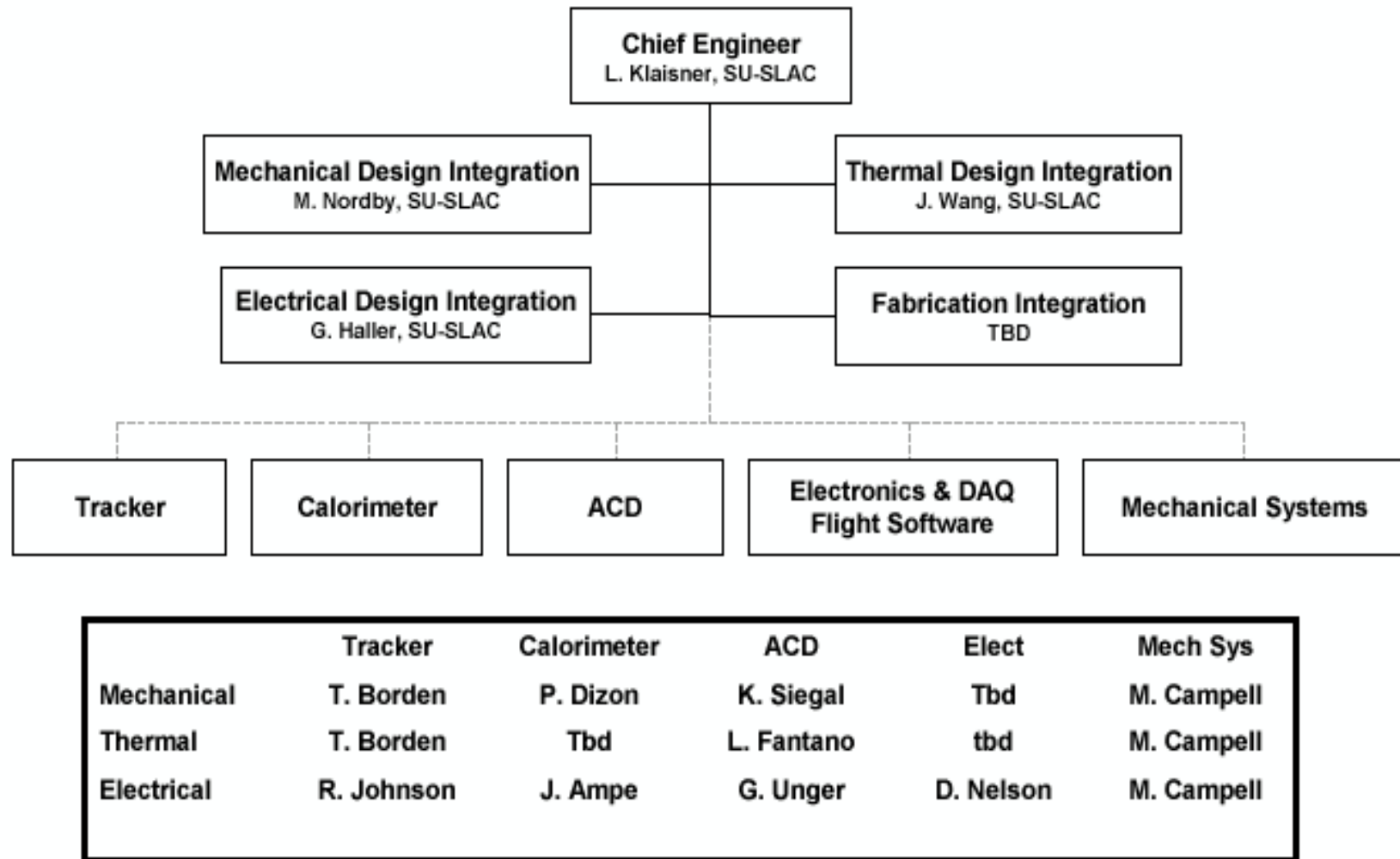
## Tech Management – Project Office 2 of 2

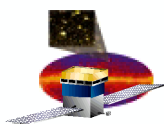
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# Tech Management – Engineering Org



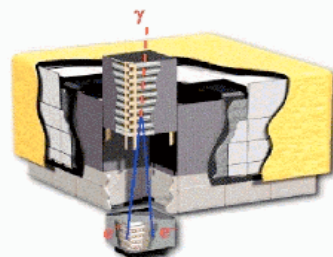


# Tech Management – Engineering Meetings



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[Tracker](#)  
[Calorimeter](#)  
[ACD](#)  
[Electronics & DAQ](#)  
[Mechanical](#)  
[Integration & Test](#)  
[Quality Assurance](#)  
[Science Analysis Software](#)  
[EPO](#)

### Instrument Design Team

[Face to Face Meeting I](#)  
[Face to Face Meeting II](#)  
[Weekly IDT Meetings](#)

### Data Analysis

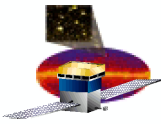
[Test Beam](#)  
[Balloon Flight](#)  
[LAT Analysis Group](#)

### LAT Team Science Working Groups

[Dark Matter](#)  
Diffuse Radiation  
Extragalactic Sources  
Galactic & Unidentified Sources  
Transients:  
Gamma-Ray Bursts  
and Solar Flares

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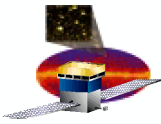
[Stanford Linear Accelerator Center](#)



# Design and System Engineering

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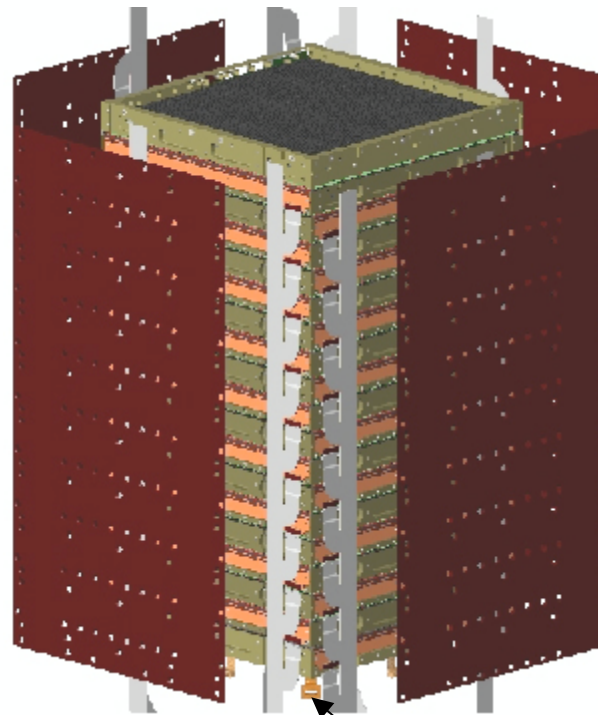
- **Design Engineering**
  - Hardware design
  - Analysis and Documentation
  - Qualification and verification
  - Manufacturing of the flight hardware
  - Alignment and Clearance
  - Coupled loads analysis
  - Design and construction of the GSE
  - Design Trades
  - Cost and Schedule
  - Traceability
- **System Engineering**
  - Requirements Analysis
    - Flowdown
    - Allocation
    - Reviews
  - Environmental Specs
  - System Control
    - Risk, Configuration and Interface Management
    - Metrics and Reviews
  - External interfaces
    - Spacecraft, Mission
    - Command and Telemetry



# Issue – Tracker Bottom Tray Design 1 of 4

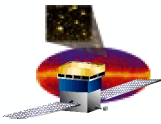
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**Tracker Module**



**Mounting Flexure**

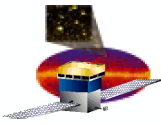




## Issue – Tracker Bottom Tray Design – 2 of 4

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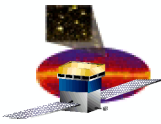
- **Issue**
  - Crack formed in the bottom tray during a vibration test of a pre-engineering model
- **Resolution**
  - Anomaly Review Team (ART) formed
    - Analyzed the failure
- **Present status**
  - Two parallel paths
    - Invar bottom tray
      - “Bullet Proof”
      - Matches thermal coefficient but is magnetic
    - Reinforce Corners
      - Requires further analysis and testing



## Issue – Tracker Bottom Tray Design 3 of 4

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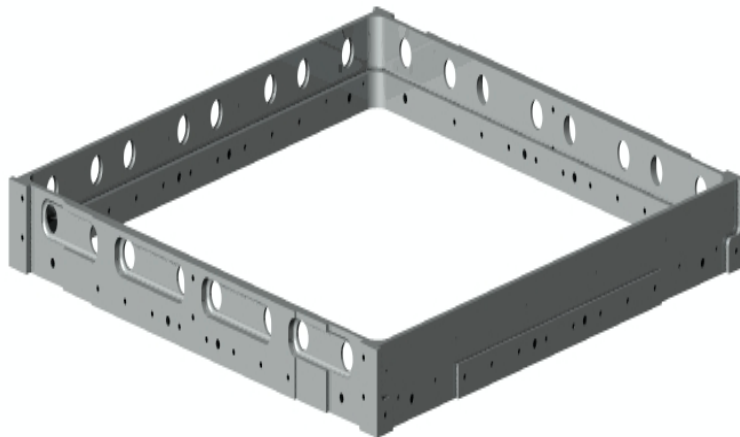
- **Schedule**
  - Resolve Invar magnetic issues by end of October '02
    - Effect on ACD phototubes
    - Effect on Spacecraft magnetometers and torque rods
  - In the meantime, check availability of alternate material for the redesign
  - Complete proposal for the design and analysis of the corner reinforcement design by end of October '02
  - In either case be ready for a design review and begin construction of the EM bottom tray by the middle of December '02 (Rest of EM proceeding independently)
  - Test a full size mechanical EM before the CDR in April '03



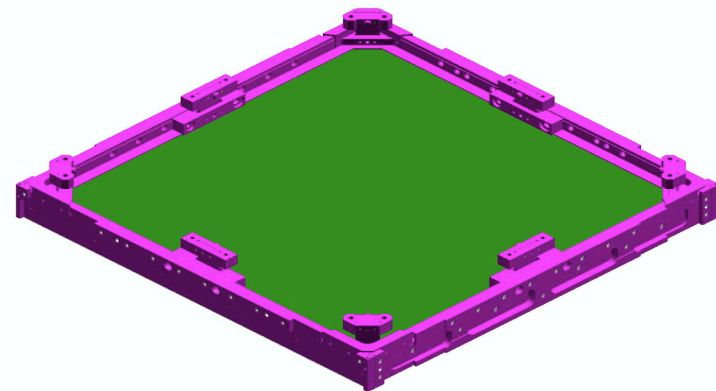
# Issue – Tracker Bottom Tray Design 4 of 4

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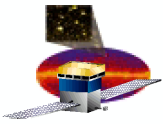
## Parallel Paths



**INVAR Bottom Tray**



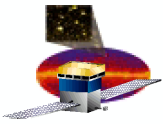
**Corner Gussets**



## Issue – Calorimeter Photodiodes 1 of 3

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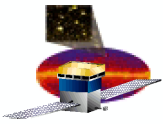
- **Issue**
  - Bond failure and cracking of the photodiode epoxy encapsulant
- **Resolution**
  - Switch to a silicone elastomer encapsulant
    - Testing commercial version
  - Hamamatsu is investigating modifying mix ratios to produce a softer epoxy
- **Present Status**
  - Elastomer commercial version received and thermal cycled successfully
  - Hamamatsu developing a timeline with early procurement of low risk items – ceramic carriers and silicon die



## Issue – Calorimeter Photodiodes 2 of 3

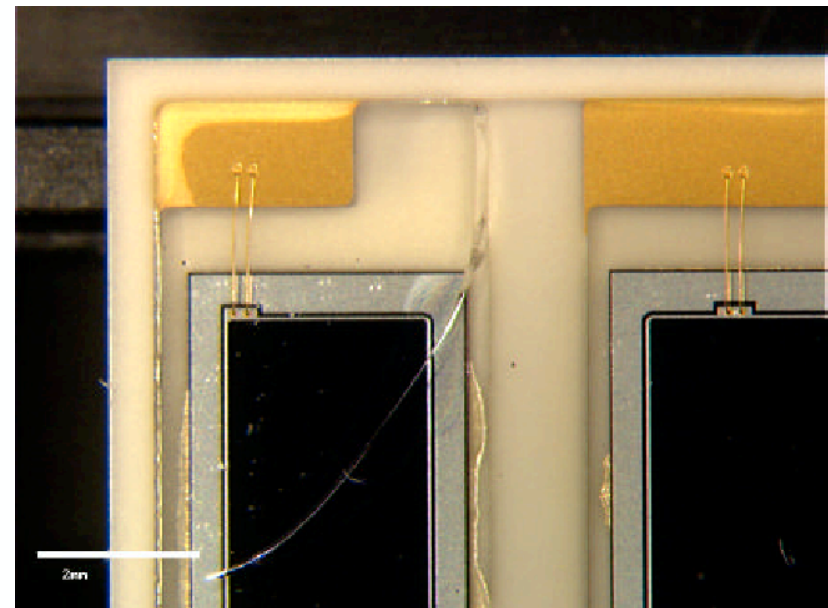
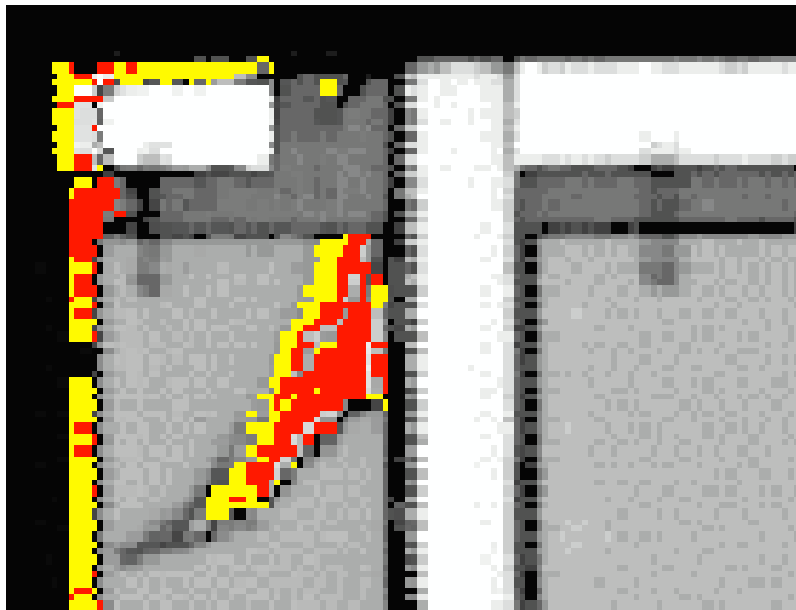
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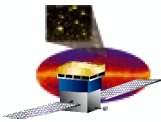
- **Schedule**
  - Plans depend on tests of elastomer units and Hamamatsu's response to a time line with early procurements
  - Still possible to meet the CDR deadline but at high risk



## Issue – Calorimeter Photodiodes 3 of 3

- Acoustic Microscopy
- Optical Image





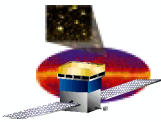
## Issue – ASICs

### •Issue

–Concern about the timely availability of final versions of the ASICS

### •Present Status and Schedule

ASIC		FH Proc	Working	Proto	EM	Notes
GTFE	TKR	Dec-02	Yes		Yes	Oscillates, noise being investigated
GTRC		Dec-02	Yes		Yes	Parity Bit and multihit TOT
GCFE	CAL	Sep-02	Yes		Yes	
GCRC		Sep-02	Yes		Yes	
GAFE	ACD	Mar-03		Nov-02		Not fully functional
GARC		Mar-03	Yes	Nov-02		
GLCC	LAT	Aug-03		Sep-02		2 more prototype cycles
GTCC		Aug-03		Nov-02		2 more prototype cycles
GCCC		Aug-03		Nov-02		2 more prototype cycles
Notes:	1	Dates are when the design is submitted for fabrication				
	2	"Working" means that the existing design could be used for flight but there are known limitations that can be overcome by another design cycle				

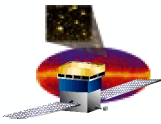


## Other Issues

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1. **Finalize Spacecraft Interface**
2. **Environmental Specification**
3. **CAL – GRID interface**
4. **Test Instrumentation**
5. **Test Plans**

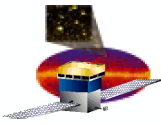




## Scope of the design effort

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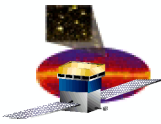
- **Expect 100% Documentation Released by LAT CDR**
  - Signed off, in Cyberdocs and under Configuration Management
  - Exceptions <10% of the total
  - Need action plan for releasing the exceptions
- **Documentation includes**
  - All Engineering Drawings
  - All Process Control Documentation
  - Manufacturing procedures
- **All Parts, Materials, and Processes must be on the LAT Approved List**



## Scope of the design effort

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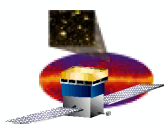
- **500 to 1000 Engineering Drawings**
  - **25 Weeks to CDR**
  - **20 to 40 Engineering Drawings Released per Week**
- **200 to 500 Manufacturing procedures**
  - **8 to 20 Procedures Released per Week**
- **Complete Drawing Tree by end of November**
  - **Determine Status and Monitor progress**
  - **Includes all electrical and mechanical drawings**
- **Initial Materials and Processes List by the end of November**
- **Develop a list of planned PRRs**
- **Need to review the Configuration Management procedure**



## EM Deliverables to I&T

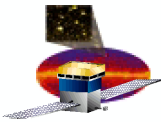
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- **ACD** – none
- **TKR** - one full size mechanical module and a four-tray module fully instrumented (1 with W, 2 No-W and 1 bottom tray) with cables and electronic readout (mini-tower), TKR lift fixture, preliminary functional test scripts
- **CAL** – fully instrumented module, preliminary functional test scripts
- **ELX/I&T** - EGSE – EM1 version
- **ME/I&T** - 1 x 4 support grid
- **SAS/I&T** - GLEAM Monte Carlo, calibration algorithms

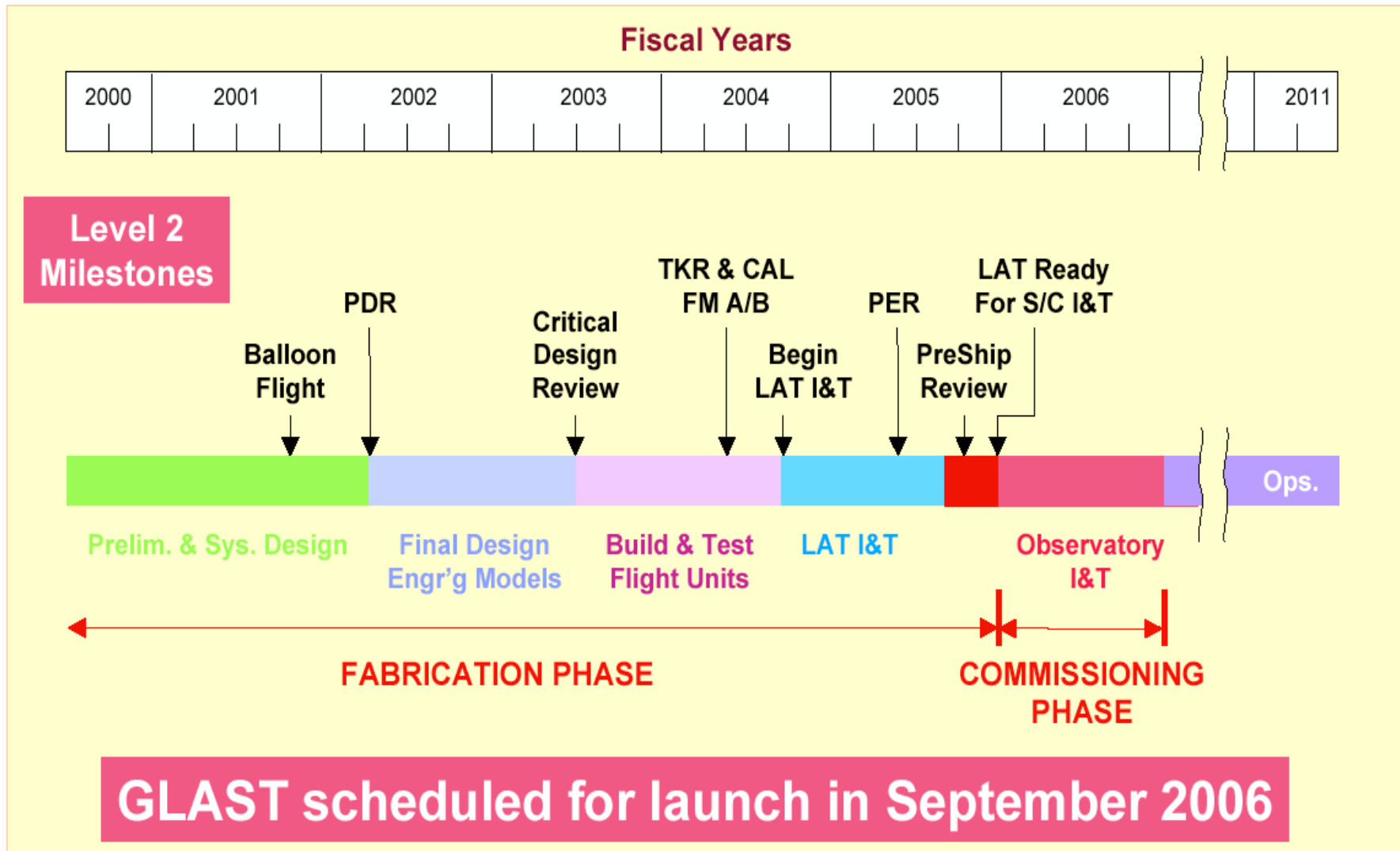


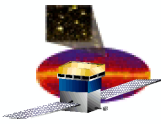
## EM Delivery Schedule

<b>Subsystem/Hardware</b>	<b>Delivery to I&amp;T (baseline dates)</b>	<b>Delivery back to subsystem</b>
<b>ME – EM/CU Grid</b>	<b>December 2, 2002</b>	<b>Not required</b>
<b>TKR – EM Mechanical Model</b>	<b>December 9, 2002</b>	<b>Not required</b>
<b>TKR – EM live mini-tower</b>	<b>February , 2003 (not in baseline)</b>	<b>Not required</b>
<b>ELX – EGSE Hardware</b>	<b>February 2003</b>	<b>Not required</b>
<b>CAL – EM Calorimeter</b>	<b>April 25, 2003</b>	<b>June 6, 2003</b>



# Schedule





# Spacecraft Vendor

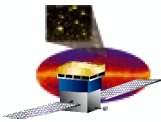
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# Summary

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- **The Management Team is in place**
- **A Spacecraft Vendor has been chosen**
- **The Project has passed the NASA Preliminary Design Review**
- **The Project has passed the DOE CD-2 Review**
- **The Project is in the detailed design and test phase**
  
- **There are significant technical challenges**
- **Most systems and procedures are not in place**
- **Cost and schedule are very tight**
- **Focusing on a successful Critical Design Review in April '03**